B. Calculation of C/[N+1] Due to Peak MSS Satellite Signal Power in the MAT Receiver

Equations 5 to 9 are applicable to the interfering signal path between the MSS satellite and the MAT receiver. It is assumed that the MSS and MAT signals have the same polarization, which is the worst-case coupling situation with regard to any MAT link.

$$D' = 42736.6[1 - 0.2951(\cos(\zeta))(\cos(\beta))]^{0.5}$$
(5)

where:

D' = slant path distance (km) from the MSS satellite to the mobile earth station receiver.

$$E' = \cos^{-1}[\{(6378 + 35880)\sin(\cos^{-1}(\cos\zeta\cos B)\}/D]$$
 (6)

where:

E' = elevation angle (degrees) from the Mat receiver to the MSS satellite.

$$A = \cos^{-1}[(\tan\zeta)\cot\{\cos^{-1}(\cos\zeta\cos\beta)\}] \tag{7}$$

where:

A = azimuth angle (degrees) from the MAT receiver to the MSS satellite.

$$C/[N+I] = C/N + 10 \log [N/\{N + 10^{-0.1}(P_t' + G_t' + -20 \log (D' \cdot f) - 32.45 + G_t')\}]$$
(8)

where:

 $G_r' = gain of the MAT receiver antenna toward the MSS satellite (dBi).$

The MAT receiver antenna patterns are given in Ref. 3 for antennas with diameters of 2.44 m and 10 m antenna. Due to apparent anomalies in the antenna gain equations and corresponding curves (Figure 1) in Ref. 3, an alternative equation was substituted for small offaxis angles ($\leq 6.35^{\circ}$ and $\leq 1.84^{\circ}$ for the 2.44m and 10m antennas, respectively) and is given in Equation 9 below. Because the center of the Earth, MAT receiver, test aircraft, and MSS satellite are located in the same plane in Geometry 1 (see Section 3), the difference in elevation angles from the MAT receiver to the aircraft and to the satellite is the off-axis angle used to determine the MAT antenna gain G_r .

$$G_{r}' = G_{r} - 2.5 \times 10^{-3} (d \cdot \theta/\lambda)^{2}$$
 (9)

where:

 θ = MAT receiver antenna off-axis angle (degrees) toward the MSS satellite.

C. Availability Calculations for Desired Signals

Ref. 3 (Section 2.3) suggests that Rayleigh fading of the desired signal in a MAT link should be considered as a near-worst-case. This type of fading occurs in links subject to persistent blockage of the direct line-of-sight signal path, as could occur with use of a single antenna that is flush-mounted off the centerline of the fuselage of the test aircraft. Measured samples of fading statistics are presented in the form of MAT transmitter antenna gain statistics in Ref. 3 (Figure 3). This distribution indicates virtually 100% probability that a "gain" of 10 dBi will not be exceeded and the 0 dBi "gain" level is not exceeded for 96.86% of the time, which yields the following near-worst-case cumulative distribution of C/[N+I] assuming a constant interference level (or C/N in the absence of interference):

$$P(C/[N+I]) = 100 e^{-3.46X}$$
 (11)

$$X = 10^{(0.1\{C/[N+I] - C/[N+I]unfaded\})}$$
(12)

where:

P(C/[N+I]) = percentage of time that the C/[N+I] level is exceeded;

C/[N+I] = carrier to noise plus interference power ratio (dB) of interest;

 $C/\{N+I\}_{unfaded}$ = unfaded carrier to noise plus interference power` ratio (dB) calculated for the Geometry under consideration.

The inherent radio path availability is determined from the above cumulative distribution of fading by substituting the unfaded desired signal level (C/N from equation 3 plus N from equation 4) for C and substituting the minimum required signal level (minimum C/N from section 2, below, plus N from equation 4) for C_{th}. The classical "fade margin" (in dB) is defined as the unfaded C/N (Equation 3) C - 12 dB (the assumed performance threshold). Although Ref. 3 does not specify any interval of time over which availability should be addressed, it is assumed that there is a certain maximum time interval during which unacceptably low availability corresponds with "loss" of test data or triggers termination of testing. For analysis purposes, this time interval is taken to be ten consecutive seconds (i.e., typical interval for data communications).

It is noted in Ref. 3 that distributions corresponding to an exponent of (-5C_{th}) have been observed. In such a MAT link, larger fade margins than are indicated by equation 11 are needed to achieve a given availability. In the alternative, some MAT links may have signal paths that are not persistently blocked, such as in cases where a single antenna is mounted on the belly of the test aircraft near the centerline of the fuselage. In such cases, the signal fading is caused mainly by surface and atmospheric multipath, wherein the Earth surface or atmospheric strata scatters the transmitted signal toward the MAT receiver and these indirect signals add at various phase offsets to the direct line-of-sight (or slightly blocked) signal. This fading, referred to as Rician fading, becomes increasingly severe as the elevation angle from the MAT receiver antenna to the test

aircraft is reduced. A certain percentage of the MAT links will exhibit Rician fading, but these are not addressed herein because Rayleigh fading is the worst case.

D. Variability of MSS Signals Due to Changes in MAT Receiver Antenna Discrimination

The two MAT operating scenarios (Geometries 1 and 2 which is defined in Section 3) occur for certain instants of time, and the calculated interference levels for those instants of time include the two worst-case local maxima over all time and test aircraft locations. Less interference will occur at all other instants of times (or aircraft locations). As shown in Table 3, the onset or falloff of interference from the peak value determined in Geometry 1 (maximum test aircraft range and worst-case MAT receiver antenna azimuth) is slow in relation to the assumed ten second intervals during which availability should be measured. Conversely, the onset or falloff of interference from the peak value determined in Geometry 2 (minimum test aircraft range and MAT receiver antenna pointed at MSS satellite) is fast in relation to the assumed ten second intervals during which availability should be measured.

Test Aircraft Velocity	Change in 10 meter MAT Antenna Discrimination in 10 Sec.											
Relative to MAT Receiver	Geometry 1	Geometry 2										
300 km/hr, radially	0.05	6.2										
300 km/hr, azimuthally	0.0	0.01										
500 km/hr, radially	0.05	16.8										
500 km/hr, azimuthally	0.0	0.02										
700 km/hr, radially	0.05	21.3										
700 km/hr, azimuthally	0.0	0.04										

Note: The indicated changes in MAT antenna discrimination are the peak values occurring during any 10 second interval that start or end with the MAT receiver antenna pointing situation of Geometries 1 or 2.

Ref. 3 does not provide details of aircraft operating areas that would enable rigorous evaluation of the statistics of interfering signal levels. However, Ref. 3 introduces an "S" parameter (Section 6), where S is the angular area centered on the MSS signal path over which the MAT receiver antenna is pointed during telemetry reception. The angular area S is applied in Ref. 3 under the assumption that the MAT receiver antenna and test aircraft flight patterns are oriented to enable direct pointing of the MAT receiver antenna at the MSS satellite (i.e., a worst-case situation). The angular area S' within which availability is measured is as follows:

$$S = \pi/4(\beta)(\Delta\alpha) \text{ steradians}$$
 (12)

$$\beta = 2\tan^{-1}[a/(d+a)] \text{ radians}$$
 (13)

$$\Delta \alpha = \tan^{-1}[(h/(d-a)) - (d-2a)/2r] - \tan^{-1}[(h/d) - (d/2r)] \text{ radians}$$
 (14)

where:

S = angular area (steradians) centered on the MSS signal path over which the MAT receiver antenna is pointed during telemetry reception during normal testing;

```
    β = elevation pointing span of MAT receiving antenna (radians);
    Δα = azimuthal pointing span of MAT receiving antenna (radians);
    h = maximum altitude (km) of test aircraft (20 km);
    d = maximum great circle distance (km) to aircraft (320 km);
    a = minimum radius (km) of flight patterns (20 km);
    r = radius (km) of the Earth (6376 km).
```

The above calculations yield the following relationship between S and the angle of arrival α (radians) of the MSS signal (elevation above horizon of the MAT receiver antenna), over angles α that occur in this analysis:

```
S = 1.557\alpha^{2.52}, for 0.11217 (6.43°) \leq \alpha \leq 1.09075 (62.5°)
= 1.9380, for \alpha > 1.09075 (62.5°)
```

The ratio of angular areas S'/S relates to the probability of interfering signal power being less than a level I' that is associated with S', where S' is determined by an off-axis angle ϕ at the MAT receiving antenna toward the MSS satellite. That is, the interfering signal level I occurs at the off-axis angle ϕ . By setting ϕ equal to the angle where interference is at the level corresponding with the permissible level, the ratio S'/S yields the probability of exceeding the permissible level of interference. If this probability is less than the unavailability allocated to interference, the interference is considered to be acceptable.

E. Variability of MSS Downlink EIRP

The standard channel used in the system now under construction by AMSC alternately operates in digital in-band signaling, data, or voice modes using fixed frame formats. In relation to transmission statistics for voice or data, the transmission statistics for the in-band signaling mode are insignificant. In the voice mode, the CODEC generates frames that are transmitted only when speech is detected, and no transmission occurs at other times during normal speech or while the user is listening. In the voice mode, the probability of frame transmission in the downlink direction is about 0.35. In the data mode, a lengthy contiguous series of frames can be transmitted and there is a downlink transmission probability of about 0.5 when operating in this mode. There are about 80 MSS downlink transmission frames during the 10 sec. time interval of concern to MAT availability, and so, the Central Limit Theorem of statistics can be applied with

respect to the three operating modes. Specifically, assuming an MSS channel has been assigned, there is a 0.35 probability of transmission for the MSS voice mode during small time intervals within the 10 sec. interval, and there is a 0.5 probability of transmission for the data mode during the entire 10 sec. interval. Only the latter statistic is considered in this study.

ATTACHMENT 2

SPREAD SHEET CALCULATIONS OF INTERFERENCE TO MAT

assumptions:

frequenc 1500 MHz

10 m Antenna 101 2.44 m Antenn Longitud Max Gai 29 Max Gai 41.2 Delta Elev MSS Parameters Aeronautical Telemetry Parameters C/N C/N+I Angle C/N C/N+I Gr Gt Т В Ν Pt' Gt' Distance Elev Azimuth Gr Longitud Latitude Altitude Range Distance Elevatio Azimuth Pι 0.177 30.5 -5.953 43.7 43.4 43.464 200 L0E+06 ** 32 37465.8 45.597 65.284 31.5 320.0 321.09 2.133 65.284 3 0 66.7 18.3 20 200 1.0E+06 ** 32 38306.5 34.637 39.316 3.331 31.5 29.8 -2.799 43.7 43.2 32,505 72.8 40 9 320.0 321.09 2.1327 39.3155 3 0 3.5 20 43.3 37 658 320.0 321.09 2.1327 37.8342 3 0 200 1.0E+06 ** 3.5 32 37891.8 39.790 37.834 1.734 31.5 30.2 -4.396 43.7 76 36.9 20 2.033 30.2 -4.097 43.7 43.3 36.634 200 1.0E+06 ** 32 37971 5 38 767 36 454 31.5 76.4 38.3 20 320.0 321.09 2.1327 36.4536 3 0 3.5 48.943 -0 901 31.5 30.7 -7 000 43 7 43.5 200 1.0E+06 ** 3.5 32 37106.5 51.075 28.375 320 0 321 09 2.1327 28.3746 3 0 85.8 30.2 20 200 1.0E+06 ** 30.7 -7.000 43.7 43.5 48.960 3 0 3.5 32 37105.4 51.092 27.001 -0.90131.5 320.0 321.09 2.1327 27.0012 86.5 30.5 20 2.1327 16.6291 3 0 200 1.0E+06 ** 3.5 32 37597.1 43.740 16.629 0.651 31.5 30.5 -5.479 43.7 43.4 41.608 90.4 38.8 20 320.0 321.09 200 1.0E+06 ** 37113.6 50.959 -7.000 43.7 43.5 48.826 32 9.795 -0 901 31.5 30.7 106.4 33.2 20 320.0 321.09 2.1327 9.7946 3 0 3.5 43.4 38.860 200 1.0E+06 ** 37799.9 40.993 18.240 1.392 31.5 30.3 -4.738 43.7 320.0 321.09 2.1327 18.2399 3 0 3.5 32 113.2 41 20 3 O 200 1.0E+06 ** 3.5 32 37231.7 49.091 23.682 -0.663 31.5 30.7 -6.793 43.7 43.5 46.958 32.9 320.0 321.09 2.1327 23.6820 114.4 20 200 1.0E+06 ** 37638.2 43.172 24.782 0.800 31.5 30.4 -5.330 43.7 43.4 41.040 116.8 37.8 20 320.0 321.09 2.1327 24.7822 3 0 3.5 37 37584,8 43.912 26.312 43.4 41,779 200 1.0E+06 ** 31.5 30.5 -5.524 43.7 321.09 2.1327 26.3121 3 0 3.5 32 0.606 117.5 36.8 20 320.0 200 1.0E+06 ** 3.5 32 37462.9 45.639 27.969 0.166 31.5 30.6 -5.96443.7 43.4 43,507 320.0 321.09 2.1327 27.9694 3 0 117.9 34.9 20 3 O 200 1.0E+06 ** 3.5 32 37445.9 45.884 30.242 0.105 31.5 30.6 -6.025 43.7 43.4 43.751 34.1 20 320.0 321.09 2.1327 30.2420 119.1 43.4 200 1.0E+06 ** 37436.1 46.027 31.958 0.070 31.5 30.6 -6.060 43.7 43.894 120 33.5 20 320.0 321.09 2.1327 31.9581 3 0 3.5 32 3 0 200 1.0E+06 ** 3.5 32 37522.7 44.783 31.806 0.382 31.5 30.5 -5.748 43.7 43.4 42.651 20 320.0 321.09 2.1327 31.8057 120.4 34.6 200 1.0E+06 ** 37796.5 41.038 32.906 31.5 30.3 -4.750 43.7 43.4 38.906 0 3.5 32 1.380 122.5 37.5 20 320.0 321.09 2.1327 32.9056 3 22 320.0 321.09 2.1327 77.1690 3 0 200 1.0E+06 ** 3.5 32 39582.8 20.680 77.169 9.423 31.5 27.0 3.293 43.7 42.1 18.547 159.7 20 200 1.0E+06 ** 32 38306.5 34.637 39.316 3.331 31.5 29.8 -2.799 43.7 43.2 32,505 72.8 40.9 20 320.0 321.09 2.1327 39.3155 3 0 3.5 43.2 32.406 200 1.0E+06 ** 31.5 29.8 -2.766 43.7 320.0 321.09 2.1327 38.7933 3 0 3.5 32 38314.8 34.538 38.793 3.364 73.1 41.2 20 0 200 1.0E+06 ** 3.5 32 38129.2 36.790 36.880 2.635 31.5 30.0 -3.495 43.7 43.3 34.657 320 0 321.09 2.1327 36.8802 3 75.3 39.9 20 1.0E+06 ** 32 38102.7 37.118 36.626 2.533 31.5 30.0 -3.597 43.7 43.3 34.985 321.09 2.1327 36.6257 0 200 3.5 75.6 39.7 20 320.0 -5.285 43.7 43.4 40.870 79 35.1 320.0 321.09 2.1327 35.0937 3 0 200 1.0E+06 ** 3.5 32 37650.5 43.003 35.094 0.845 31.5 30.4 20 1.0E+06 ** 200 32 37125.3 50.771 39.728 -0.90131.5 30.7 -7.000 43.7 43.5 48.639 80.2 27.2 20 320.0 321.09 2.1327 39.7278 35 30.7 -7.000 43.7 43.5 48.965 200 1.0E+06 ** 37105.1 51.097 39.868 -0.901 31.5 80.3 26.9 20 320.0 321.09 2.1327 39.8683 3 0 3.5 32 1.0E+06 ** 321.09 2.1327 25.5302 3 200 32 37821.0 40.715 25.530 1.470 31.5 30.3 -4.66043.7 43.4 38.582 84 39.8 20 320 0 321.09 2.1327 27.9724 3 0 1.0E+06 ** 32 37382.9 46.805 27.972 -0.12131.5 30.6 -6.25143.7 43.5 44.672 84.5 33.9 20 320.0 3.5 84.9 44.6 20 320.0 321.09 2.1327 22.3462 3 0 1.0E+06 ** 32 38179.9 36.167 22.346 2.832 31.5 29.9 -3.29843.7 43.3 34.035 2.1327 28.0256 3 0 200 1.0E+06 ** 32 37190.1 49.740 28.026 -0.81231.5 30.7 -6.942 43.7 43.5 47.607 85.5 31.4 20 320.0 321.09 1.0E+06 ** 3.5 32 38371.0 33.870 9.970 3.591 31.5 29.7 -2.53943.7 43.2 31.737 93.5 48.5 20 320.0 321.09 2.1327 9.9697 3 93.5 45.5 20 320.0 321.09 2.1327 10.4580 3 1.0E+06 ** 32 38103.5 37.108 10.458 2.536 31.5 30.0 -3.594 43.7 43.3 34.975

1. INTRODUCTION

1.1 Background

For over fifteen years, the 1435-1535 MHz band has been allocated in the US to the Mobile Aeronautical Telemetry (MAT) service. MAT involves transmissions from test aircraft through low gain antennas to receiving ground stations that use high-gain tracking antennas. The 1979 World Administrative Radio Conference (WARC-79) allocated the 1530-1535 MHz segment of the subject band to the maritime Mobile-Satellite Service (MSS) in the space-to-Earth direction on a shared basis with MAT, which is accommodated internationally under ITU Radio Regulation (RR) No. 723. Subsequently, this shared allocation was adopted in the US with a provision that the MAT allocation be reduced to secondary status in 1990 (US 78). NTIA studied this frequency sharing situation and recommended that MAT transmitters (e.g., test aircraft) located in coastal areas avoid using the same frequencies as maritime MSS downlinks in order to protect receiving ship earth stations (Ref. 1a). Another NTIA study of sharing at 1530-1535 MHz between MSS downlinks and MAT systems concluded that "[a] signal from the maritime mobile-satellite is not strong enough to interfere with the telemetry receiver on earth" (Ref. 1b). However, that conclusion assumes MSS downlink Power Flux Density (PFD) levels of the order of -144 dBW/m²/4 kHz, which are less than the levels that currently are achievable and desirable in domestic MSS systems.2

More recently, in US preparations for the 1992 WARC, MSS proponents evaluated the potential for sharing between MSS downlinks and MAT systems in the 1435-1530 MHz band. These studies concluded that sharing is feasible and that a 10 MHz or larger segment of that band would be among the most preferred new allocations for the MSS. Based on US and other proposals, WARC-92 allocated the 1525-1530 MHz band to MSS (space-to-Earth) on a worldwide basis. In addition, based on proposals from Canada and Brazil, WARC-92 allocated the 1492-1525 MHz for MSS (space-to-Earth) transmissions in the Americas (ITU Region 2). However, despite the demand for MSS in the US, the US declined to adopt the latter new MSS allocation (RR No. 722C).

After WARC-92, AMSC's interests in using these MSS allocations in the US were documented (Ref. 2) and distributed to certain MAT authorities in order to facilitate further discussions of the subject frequency sharing situation. In the time since WARC-92, however, MAT authorities have had to focus on the development of a coordination trigger for protection of

That NTIA study did not address potential interference from MAT systems to aircraft earth station operations that were (and remain) permissible in the 1525-1535 MHz band under RR Article 48. In addition, subsequent to the NTIA Report, the 1987 WARC for Mobile Telecommunications expanded the MSS allocations in the 1533-1535 MHz segment to include land MSS operations. Thus, all types of mobile earth stations could experience interference from MAT systems in the 1530-1535 MHz band. However, other studies have indicated that the means for preventing interference to MAT also would prevent interference to MSS.

The ship earth stations in operation at the time of the NTIA studies used antennas with diameters of 1.0 m to 1.2 m with gain levels greater than 20 dBi in the 1530-1535 MHz band. Earth stations for use on land, boats and aircraft are necessarily smaller and have substantially lower gain levels that require higher PFD levels.

																							•
93.7	32.5	20	320.0	321.09	2.1327	13.4101	3	0	200	1.0E+06	**	3.5	32	37088.2	51.373	13.410	-0.901	31.5	30.7	-7.000	43.7	43.5	49.240
95.9	36.2	20	320.0	321.09	2.1327	8.5931	3	0	200	1.0E+06	**	3.5	32	37325.7	47.657	8.593	-0.326	31.5	30.6	-6.456	43.7	43.5	45.525
97.3	37.6	20	320.0	321.09	2.1327	6.0500	3	0	200	1.0E+06	**	3.5	32	37420.3	46.256	6.050	0.013	31.5	30.6	-6.117	43.7	43.4	44.123
97.5	32.8	20	320.0	321.09	2.1327	6.4418	3	0	200	1.0E+06	**	3.5	32	37070.4	51.664	6.442	-0.901	31.5	30.7	-7.000	43.7	43.5	49.531
97.8	32.7	20	320.0	321.09	2.1327	5.9084	3	0	200	1.0E+06	**	3.5	32	37061.7	51.807	5.908	-0.901	31.5	30.7	-7.000	43.7	43.5	49.675
105.9	37.4	20	320.0	321.09	2.1327	8.0341	3	0	200	1.0E+06	**	3.5	32	37414.1	46.348	8.034	-0.009	31.5	30.6	-6.139	43.7	43.4	44.215
106.1	32.9	20	320.0	321.09	2.1327	9.3308	3	0	200	1.0E+06	**	3.5	32	37090.1	51.342	9.331	-0.901	31.5	30.7	-7.000	43.7	43.5	49.209
106.2	32.1	20	320.0	321.09	2.1327	9.7182	3	0	200	1.0E+06	**	3.5	32	37036.7	52.222	9.718	-0.901	31.5	30.7	-7.000	43.7	43.5	50.089
106.3	39.2	20	320.0	321.09	2.1327	8.3501	3	0	200	1.0E+06	**	3.5	32	37557.0	44.300	8.350	0.505	31.5	30.5	-5.625	43.7	43.4	42.168
106.5	48.4	20	320.0	321.09	2.1327	7.3372	3	0	200	1.0E+06	**	3.5	32	38343.5	34.196	7.337	3.480	31.5	29.7	-2.650	43.7	43.2	32.063
110.4	31.6	20	320.0	321.09	2.1327	17.5335	3	0	200	1.0E+06	**	3.5	32	37061.0	51.818	17.533	-0.901	31.5	30.7	-7.000	43.7	43.5	49.685
110.7	33.5	20	320.0	321.09	2.1327	17.2076	3	0	200	1.0E+06	**	3.5	32	37194.1	49.678	17.208	-0.798	31.5	30.7	-6.928	43.7	43.5	47.545
113.4	40.2	20	320.0	321.09	2.1327	18.8105	3	0	200	1.0E+06	**	3.5	32	37740.6	41.782	18.811	1.174	31.5	30.4	-4.956	43.7	43.4	39.649
114.6	32.5	20	320.0	321.09	2.1327	24.2402	3	0	200	1.0E+06	**	3.5	32	37210.1	49.426	24.240	-0.740	31.5	30.7	-6.870	43.7	43.5	47.294
115.1	39.3	20	320.0	321.09	2.1327	21.6320	3	0		1.0E+06		3.5	32	37708.1	42.219	21.632		31.5	30.4	-5.075	43.7	43.4	40.087
115.3	36.2	20	320.0	321.09	2.1327	23.3443	3	0		1.0E+06		3.5	32	37482.8			0.238	31.5	30.5	-5.892	43.7	43.4	43.220
117.2	32.8	20	320.0	321.09	2.1327	28.2054	3	0		1.0E+06		3.5	32	37300.9			-0.415	31.5	30.7	-6.545	43.7	43.5	45.898
117.3	33.1	20	320.0	321.09	2.1327	28.1677	3	0		1.0E+06		3.5	32	37323.7				31.5	30.6	-6.463	43.7	43.5	45.555
118.2	35.1	20	320.0	321.09	2.1327	28.2956	3	0	200	1.0E+06	**	3.5	32	37485.7			0.248	31.5	30.5	-5.882	43.7	43.4	43.179
118.2	34.4	20	320.0	321.09	2.1327	28.7188	3	0		1.0E+06		3.5	32	37437.8				31.5	30.6	-6.054	43.7	43.4	43.869
118.2	33.8	20	320.0	321.09		29.0938	3	0		1.0E+06		3.5	32	37397.4			-0.069	31.5	30.6	-6.199	43.7	43.5	44.459
118.3	33.9	20	320.0	321.09		29.1806	3	0		1.0E+06		3.5	32	37407.2			-0.034	31.5	30.6	-6.164	43.7	43.4	44.315
118.4	34	20	320.0	321.09	2.1327	29.2670	3	0		1.0E+06		3.5	32	37417.0			0.001	31.5	30.6	-6.129	43.7	43.4	44.172
118.4	37.4	20	320.0	321.09		27.2919	3	0	-	1.0E+06		3.5	32	37654.2			0.858	31.5	30.4	-5.272	43.7	43.4	40.820
120.1	34.6	20	320.0	321.09		31.3756	3	0		1.0E+06		3.5	32	37512.5				31.5	30.5	-5.785	43.7	43.4	42.796
120.2	39.2	20	320.0	321.09		28.8540	3	0		1.0E+06		3.5	32	37841.4				31.5	30.3	-4.584	43.7	43.4	38.314
121.2	37.4	20	320.0	321.09	2.1327	31.2061	3	0		1.0E+06		3.5	32	37743.6				31.5	30.3	-4.945	43.7	43.4	39.609
122.3	47.5	20	320.0	321.09	2.1327	27.8706	3	0	200	1.0E+06	••	3.5	32	38563.8	31.626	27.871	4.387	31.5	29.4	-1.743	43.7	43.1	29.494

C/N+I Ratios with Aero @ max altitude and same elevation angle as MSS

Geometry 2

Macro for receiver gain not used; max revr gain used
assumptions:

frequenc 1500 MHz

	frequenc	1500	MHz								Longitud	101			ŀ		2 44	m Antenn		10	m Antenna	
											Longituu	101					Max Gai	29	1	Max Gai	41.2	
_											MSS Paran	atam					May Out		•	VIAN ONI	71.2	Delta Ele
	itical Tele			P1 4'	4 ' 4	-	C.	т	ь			Gt'	Distance	Elev	Azimuth	Gr	C/N	C/N+I	Gr	C/N	C/N+I	Angle
Longitud	Latitude	Altitude	Distanc	Eievatio	Azimuth	Pt	Gt	T	В	N	Pt'	Gt	Distance	Elev	Azimuui	Gi	C/N	C/NTI	Gi	C/N	CHYI	Angle
66.7	18.3	20	27.952	45.597	65.284	3	0	200	1.0E+06	**	3.5	32	37465.8	45.5965	65.2836	29	52.7	30.0	41.2	64.9	30.0	0
72.8	40.9	20	35.073	34.637	39.316	3	0	200	1.0E+06	**	3.5	32	38306.5	34.6373	39.3155	29	50.7	28.2	41.2	62.9	28.3	0
76	36.9	20	31.181	39.790	37.834	3	0	200	1.0E+06	**	3.5	32	37891.8	39.7902	37.8342	29	51.7	29.2	41.2	63.9	29.2	0
76.4	38.3	20	31.864	38.767	36.454	3	0	200	1.0E+06	**	3.5	32	37971.5	38.7671	36.4536	29	51.6	29.0	41.2	63.8	29.0	0
85.8	30.2	20	25.682	51.075	28.375	3	0	200	1.0E+06	**	3.5	32	37106.5	51.0753	28.3746	29	53.4	30.7	41.2	65.6	30.7	
86.5	30.5	20	25.676	51.092	27.001	3	0	200	1.0E+06	**	3.5	32	37105.4	51.0923	27.0012	29	53.4	30.7	41.2	65.6	30.7	0
90.4	38.8	20	28.878	43.740	16.629	3	0	200	1.0E+06	**	3.5	32	37597.1	43.7403	16.6291	29	52.4	29.8	41.2	64.6	29.8	0
106.4	33.2	20	25.724	50.959	9.795	3	0	200	1.0E+06	**	3.5	32	37113.6	50.9591	9.79464	29	53.4	30.7	41.2	65.6	30.7	0
113.2	41	20	30.427	40.993	18.240	3	0	200	1.0E+06	**	3.5	32	37799.9	40.9925	18.2399	29	52.0	29.4	41.2	64.2	29.4	0
114.4	32.9	20	26.433	49.091	23.682	3	0	200	1.0E+06	**	3.5	32	37231.7	49.0905	23.682	29	53.2	30.5	41.2	65.4	30.5	0
116.8	37.8	20	29.180	43.172	24.782	3	0	200	1.0E+06	**	3.5	32	37638.2	43.1722	24.7822	29	52.3	29.7	41.2	64.5	29.7	\ o
117.5	36.8	20	28.789	43.912	26.312	3	0	200	1.0E+06	**	3.5	32	37584.8	43.9117	26.3121	29	52.4	29.8	41.2	64.6	29.8	0
117.9	34.9	20	27.932	45.639	27.969	3	0	200	1.0E+06	**	3.5	32	37462.9	45.6394	27.9694	29	52.7	30.0	41.2	64.9	30.0	0
119.1	34.1	20	27,817	45.884	30.242	3	0	200	1.0E+06	**	3.5	32	37445.9	45.8839	30.242	29	52.7	30.1	41.2	64.9	30.1	0
120	33.5	20	27,750	46.027	31.958	3	0	200	1.0E+06	**	3.5	32	37436.1	46.027	31.9581	29	52.8	30.1	41.2	65.0	30.1	0
120.4	34.6	20	28.347	44.783	31.806	3	0	200	1.0E+06	**	3.5	32	37522.7	44.7835	31.8057	29	52.6	29.9	41.2	64.8	29.9	0
122.5	37.5	20	30.399	41.038	32.906	3	0	200	1.0E+06	**	3.5	32	37796.5	41.0382	32.9056	29	52.0	29.4	41.2	64.2	29.4	0
159.7	22	20	56.026	20.680	77.169	3	0	200	1.0E+06	**	3.5	32	39582.8	20.6797	77.169	29	46.7	24.5	41.2	58.9	24.5	0
72.8	40.9	20	35.073	34.637	39.316	3	0	200	1.0E+06	**	3.5	32	38306.5	34.6373	39.3155	29	50.7	28.2	41.2	62.9	28.3	0
73.1	41.2	20	35.160	34.538	38.793	3	0	200	1.0E+06	**	3.5	32	38314.8	34.5383	38.7933	29	50.7	28.2	41.2	62.9	28.2	0
75.3	39.9	20	33.303	36.790	36.880	3	0	200	1.0E+06	**	3.5	32	38129.2	36.7899	36.8802	29	51.2	28.7	41.2	63.4	28.7	0
75.6	39.7	20	33.052	37.118	36.626	3	0	200	1.0E+06	**	3.5	32	38102.7	37.118	36.6257	29	51.2	28.7	41.2	63.4	28.7	0
79	35.1	20	29.272	43.003	35.094	3	0	200	1.0E+06	**	3.5	32	37650.5	43.0026	35.0937	29	52.3	29.7	41.2	64.5	29.7	0
80.2	27.2			50.771		3	0	200	1.0E+06	**	3.5	32	37125.3	50.7715	39.7278	29	53.4	30.6	41.2	65.6	30.7	0
80.3	26.9			51.097		3	0	200	1.0E+06	**	3.5	32	37105.1	51.0973	39.8683	29	53.4	30.7	41.2	65.6	30.7	0
84	39.8	20	30.596	40.715	25.530	3	0	200	1.0E+06	••	3.5	32	37821	40.715	25.5302	29	51.9	29.3	41.2	64.1	29.3	0
• •															•			•				•

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84.5	33.9	20 27.396	46.805	27.972	3	0	200	1.0E+06	**	3.5	32	37382.9	46.8051	27.9724	29	52.9	30.2	41.2	65.1	30.2	0
84.9	44.6	20 33.79	36.167	22.346	3	0	200	1.0E+06	**	3.5	32	38179.9	36.1674	22.3462	29	51.0	28.5	41.2	63.2	28.6	0
85.5	31.4	20 26.179	49.740	28.026	3	0	200	1.0E+06	**	3.5	32	37190.1	49.7399	28.0256	29	53.3	30.5	41.2	65.5	30.5	0
93.5	48.5	20 35.763	33.870	9.970	3	0	200	1.0E+06	**	3.5	32	38371	33.8698	9.96968	29	50.6	28.1	41.2	62.8	28.1	0
93.5	45.5	20 33.060	37.108	10.458	3	0	200	1.0E+06	**	3.5	32	38103.5	37.1081	10.458	29	51.2	28.7	41.2	63.4	28.7	0
93.7	32.5	20 25.573	51.373	13.410	3	0	200	1.0E+06	**	3.5	32	37088.2	51.3726	13.4101	29	53.5	30.7	41.2	65.7	30.7	0
95.9	36.2	20 27.024	47.657	8.593	3	0	200	1.0E+06	**	3.5	32	37325.7	47.6573	8.59307	29	53.0	30.3	41.2	65.2	30.3	0
97.3	37.6	20 27.64	46.256	6.050	3	0	200	1.0E+06	••	3.5	32	37420.3	46.2559	6.04999	29	52.8	30.1	41.2	65.0	30.1	0
97.5	32.8	20 25.473	3 51.664	6.442	3	0	200	1.0E+06	**	3.5	32	37070.4	51.6635	6.44181	29	53.5	30.7	41.2	65.7	30.8	0
97.8	32.7	20 25.423	3 51.807	5.908	3	0	200	1.0E+06	**	3.5	32	37061.7	51.8072	5.90842	29	53.5	30.8	41.2	65.7	30.8	0
105.9	37.4	20 27.603	46.348	8.034	3	0	200	1.0E+06	**	3.5	32	37414.1	46.3477	8.03414	29	52.8	30.1	41.2	65.0	30.1	0
106.1	32.9	20 25.586	5 51.342	9.331	3	0	200	1.0E+06	**	3.5	32	37090.1	51.3419	9.33075	29	53.5	30.7	41.2	65.7	30.7	0
106.2	32.1	20 25.280	52.222	9.718	3	0	200	1.0E+06	••	3.5	32	37036.7	52.2216	9.71818	29	53.6	30.8	41.2	65.8	30.8	0
106.3	39.2	20 28.589	44.300	8.350	3	0	200	1.0E+06	**	3.5	32	37557	44.3005	8.35006	29	52.5	29.8	41.2	64.7	29.9	0
106.5	48.4	20 35.460	34.196	7.337	3	0	200	1.0E+06	**	3.5	32	38343.5	34.196	7.33723	29	50.6	28.2	41.2	62.8	28.2	0
110.4	31.6	20 25.419	51.818	17.533	3	0	200	1.0E+06	**	3.5	32	37061	51.8182	17.5335	29	53.5	30.8	41.2	65.7	30.8	0
110.7	33.5	20 26.203	49.678	17.208	3	0	200	1.0E+06	**	3.5	32		49.6781		29	53.3	30.5	41.2	65.5	30.5	0
113.4	40.2	20 29.95	3 41.782	18.811	3	0	200	1.0E+06	**	3.5	32		41.7821		29	52.1	29.5	41.2	64.3	29.5	0
114.6	32.5	20 26.30	49.426	24.240	3	0		1.0E+06		3.5	32		49.4263		29	53.2	30.5	41.2	65.4	30.5	0
115.1	39.3	20 29.707	7 42.219	21.632	3	0		1.0E+06		3.5	32		42.2193		29	52.2	29.5	41.2	64.4	29.6	0
115.3	36.2	20 28.069	45.353	23.344	3	0		1.0E+06		3.5	32		45.3525		29	52.7	30.0	41.2	64.9	30.0	0
117.2	32.8	20 26.866	48.030	28.205	3	0		1.0E+06		3.5	32		48.0304		29	53.0	30.3	41.2	65.2	30.3	0
117.3	33.1	20 27.011	47.688	28.168	3	0		1.0E+06		3.5	32		47.6878		29	53.0	30.3	41.2	65.2	30.3	0
118.2	35.1	20 28.089	45.312	28.296	3	0		1.0E+06		3.5	32		45.3117	_	29	52.6	30.0	41.2	64.8	30.0	0
118.2	34.4	20 27.762	46.002	28.719	3	0	_	1.0E+06		3.5	32		46.0017		29	52.8	30.1	41.2	65.0	30.1	0
118.2	33.8	20 27.492	46.592	29.094	3	0		1.0E+06		3.5	32		46.5917		29	52.8	30.1	41.2	65.0	30.2	0
118.3	33.9	20 27.557	7 46.448	29.181	3	0		1.0E+06		3.5	32		46.4481		29	52.8	30.1	41.2	65.0	30.2	0
118.4	34	20 27.622			3	0		1.0E+06		3.5	32		46.3046		29	52.8	30.1	41.2	65.0	30.1	0
118.4	37.4	20 29.299			3	0		1.0E+06		3.5	32		42.9523		29	52.3	29.7	41.2	64.5	29.7	0
120.1	34.6	20 28.275			3	0		1.0E+06		3.5	32		44.9284		29	52.6	29.9	41.2	64.8	30.0	0
120.2	39.2	20 30.763			3	0		1.0E+06		3.5	32		40.4463		29	51.9	29.3	41.2	64.1	29.3	0
121.2	37.4	20 29.982			3	0		1.0E+06		3.5	32	37743.6			29	52.1	29.5	41.2	64.3	29.5	0
122.3	47.5	20 37.985	31.626	27.871	3	0	200	1.0E+06	**	3.5	32	38563.8	31.6264	27.8706	29	50.0	27.6	41.2	62.2	27.6	ļ o

CERTIFICATE OF SERVICE

I, Leslie Anne Byers, hereby certify that I have this 11th day of May, 1993, served a copy of the foregoing "Comments of American Mobile Satellite Corporation" by hand delivery to the following:

Mr. Norbert Schroeder
Program Manager, Spectrum Openness
National Telecommunications and
Information Administration
Room 4092, U.S. Department of Commerce
14th Street and Constitution Avenue, N.W.
Washington, D.C. 20230

Leslie Anne Byers